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Design, Implementation and Evaluation of a Redundancy Management System for Fault-Tolerant Wireless Devices in Harsh **Environments**

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Abstract

Wireless sensor networks (WSNs), when deployed in harsh environments, can fail prematurely due to elevated rates of component failures. To counteract this problem, fault-tolerant techniques, such as redundancy, may be used. A redundant design requires a management system. Built-in tests (BITs) are one of the most commonly used approaches for managing redundancy, but it suffers from issues such as imperfect fault coverage and common-cause failures (CCFs). In this work, a BIT based redundancy management system has been designed that makes use of a supervisory unit and a modular architecture to address issues with imperfect fault coverage and CCFs. The design has been implemented in prototype WSN devices and evaluated through reliability analysis, fault injection testing and industrial test deployments. The evaluation results have demonstrated the fault-tolerant capabilities of the proposed system design.

Keywords

Wireless Sensor Network (WSN), Fault-Tolerant, Built-in Tests (BITs), Redundancy Management System, Common-Cause Failure (CCF), Fault Coverage.

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 $R(n,p(t),c,\beta,\lambda)$ System reliability with imperfect fault coverage and CCFs

R(n,p(t), $c^{\prime}(\alpha),\beta,\lambda,\beta_{s},\gamma)$ System reliability with the supervisory unit, imperfect fault coverage and **CCFs**

List of Abbreviations

WSN Wireless sensor network

Chapter 1

1 Introduction

Wireless sensor networks (WSNs) have been gaining popularity in industrial and other monitoring applications due to their lower costs and an increased mobility, as compared to their wired counterparts. Wireless based systems also provide ease of deployment as they require little to no cabling, something that can be very expensive in some industrial plants. For example, retrofitted cables in nuclear power plants (NPPs) can introduce a cost of \$2000 per foot over their lifetime [1]. Under the right setting, wireless systems can improve health and safety, increase production and help reduce operating costs in industry [2]. WSNs have been used in industry to monitor temperature, pressure, liquid levels, equipment condition and motor vibrations, as well as ambient radiation levels (in NPPs) [3]. Several standards for industrial WSNs have also been developed such as WirelessHART, Zigbee and ISA 100.11a [4].

In a WSN, sensor nodes are autonomous devices that are equipped with sensors to measure certain physical or environment variables, as well as wireless transceivers to communicate data [5]. In industrial WSNs, devices that collect data are sometimes referred to as field devices, whereas devices that extend wireless communication range for data forwarding are called router devices [6]. The collected sensor information is transmitted to prespecified collection points, called the sink or gateway, for further analysis by the end users. A conceptual illustration of a WSN is shown in [Figure 1.1.](#page-15-1)

Figure 1.1: Topology of a WSN.

1.1 WSNs in Harsh Environments

Although WSNs have been used for various industrial monitoring tasks, the performance of a WSN system can be compromised if the deployment environment becomes harsh. The harsh environment can be characterized by elevated temperatures and/or radiation (in nuclear applications) [7] [8]. These harsh environments may result from industrial accidents (either natural or man-made) or may be inherent to the location of deployment. Harsh environments can lead to increased WSN device component failure rates that can potentially cause the entire system to fail prematurely. For example, elevated environmental conditions (such as temperature and radiation) can increase the chance of electronic component faults [9] [10]. Other issues that may arise include electromagnetic interference from machinery [7] and partial blocking of certain communication paths due to people and mobile equipment [11].

WSNs are deployed to accomplish a specific mission over a certain period of time, often in remote or hard to access locations. It is imperative that such a system works reliably over the entire duration of the mission. For example, a WSN mission could be to monitor an accident environment or adverse condition for the duration of the event. Alternatively, the mission could be to monitor a plant process during normal operation between two consecutive maintenance intervals. It should be noted that in most deployments it may not be feasible to replace or repair the WSN devices during the mission time. The failure of a WSN based monitoring system can potentially result in an information blackout that can negatively affect plant operations, or hinder mitigation activities if the WSN is used to monitor adverse plant conditions.

It is noted that the existing WSN systems deployed in relatively harsh environment typically use some kind of environment casing or shielding to protect components, and also employ some methodology to achieve system level fault-tolerance. System level faulttolerance can be achieved by forming redundant communication paths in the network (through strategic or dense node deployment) to automatically route information when some nodes fail. In certain applications, however, it might not be feasible to deploy a large number of devices to attain system level fault-tolerance, for example, in a NPP, as it's safety instruments may be sensitive to the EMI from the devices [1]. Instead, device level fault-tolerance can help enhance the overall WSN system reliability under harsh environments and can provide fault-tolerance to a system when system level fault-tolerance may not be practical.

1.2 Redundant WSN Design

As mentioned, harsh environments can increase the rate of component failures in a WSN device. Therefore, it is feasible to make a device fault-tolerant so that it can operate even if some of its components may have failed. Fault-tolerant device design is the core objective of this work. The effect of a fault-tolerant design can be expressed in terms of reliability, which is defined as the probability that a system will be operational during some specified mission time. It is noted that reliability is one of the most common ways to express a system's fault-tolerance ability [12].

A system's fault-tolerance can potentially be improved by incorporating redundancy in the design [13]. Redundancy is the act of replicating critical components in a system, such that

some components can remain as backups and assume operation only if the primary component fails [13].

In a redundant design, a redundancy management system is tasked with detecting and identifying faulty components, as well as reconfiguring the system when a fault is detected. Most of the existing redundancy management systems fall into one of the following approaches: 1) built-in tests (BITs) or 2) voting logic [14].

BITs detect and identify faults by completing a series of in-field tests for each individual component. These tests could be realized as supplementary hardware or as a softwarebased diagnostics algorithm within a component's existing logic. In WSN devices, both hardware and software BITs can be implemented to aid in the detection and recovery from faults [15]. When a fault is detected, the faulty component can be isolated. A backup component can then assume operation so that a device can continue to function. [Figure 1.2](#page-17-0) illustrates a redundancy management system using the BIT approach. In [Figure 1.2,](#page-17-0) the redundancy management system consists of the BITs and the isolation mechanism.

Figure 1.2: BIT approach for redundant components.

The second redundancy management system approach utilizes a voting element to discern faulty components from correctly working ones. A replicated set of components operate simultaneously and feed their outputs into a voting unit. Through some predetermined

voting criterion (such as majority voting or middle value selection) faulty components can then be identified [14]. [Figure 1.3](#page-18-0) illustrates a redundancy management system using the voting logic approach.

Figure 1.3: Voting logic approach for redundant components.

Each of the two redundancy management approaches comes with their own strengths and weaknesses. For instance, the BIT approach can be implemented with as little as two replicated components. It operates on a *1-out-of-n* basis, meaning that only 1 replicated component must be correctly working for the management system to work [14]. The BIT approach is more suitable for applications that have limited resources. In contrast, the voting logic approach typically requires a minimum of 3 or more replicated components and usually operates on a *2-out-of-n* basis [14]. Since voting logic requires a higher base number of replicated components and has a lower bound on the number of operational components to successfully identify faults, this approach favours applications that are repairable, i.e., can undergo maintenance during the mission times [16].

In this work, the BIT approach has been used to manage the redundancy of a fault-tolerant WSN design. The BIT approach has been chosen as it usually requires fewer resources, which is typically one of the requirements for a WSN system.

1.3 Issues with Built-in Tests

Although BITs may be better suited for resource constrained WSNs, some of its own drawbacks can potentially counteract this approach's overall reliability improvement. One of these drawbacks is imperfect fault coverage. Fault coverage is the ability of a system to correctly detect and identify a faulty component [14]. Imperfect fault coverage results if certain faults cannot be detected, which can then lead to a complete device failure. For example, if an undetected fault has occurred in a redundant system, then that fault will not be mitigated by the redundancy management system. It can be assumed that unmitigated faults result in a device failure (either directly or indirectly by causing additional faults in the system) regardless of whether backup components are available.

Another issue with the BIT approach (and redundancy management systems in general) is the risk of common-cause failures (CCFs) [17]. The elements used in a redundancy management system to detect, identify and reconfigure faulty components is also susceptible to failures. Failure of an element of the redundancy management system could trigger a complete system failure.

To design a fault-tolerant WSN using the BIT approach, both imperfect fault coverage and CCFs impacts must be effectively addressed.

1.4 Research Objectives, Scope and Methods

1.4.1 Research Objectives

The WSN system, proposed in this work, is assumed to be deployed to perform a monitoring task during certain critical missions in a harsh environment. Repairing and replacing system devices during the mission time is not feasible. Furthermore, the proposed system is particularly suitable for applications where the deployment of a large number of devices to achieve system level fault-tolerance is not practical.

The objectives of this research are to:

• design a fault-tolerant WSN device that uses the BIT-based redundancy management system approach.

- implement the redundancy management system in prototype WSN devices.
- evaluate the fault-tolerant performance of the redundancy management system.

Specifically, a redundancy management system has been designed that makes use of a supervisory unit, fault detection hardware and a modular design to address the problem of imperfect fault coverage and CCFs. This design has been realized in prototype WSN devices and their performance has been evaluated in an experimental setting.

1.4.2 Research Scope

The proposed design considers fault-tolerant WSN devices with component level redundancy using the BIT approach. The WSN is assumed to be non-repairable during its mission time. The design has been realized in a prototype WSN system that is then evaluated based on assumed and estimated reliability model parameters under simulated harsh environment conditions. Elevated levels of temperature and radiation have been considered during the system evaluation. These elevated levels are assumed to not be severe enough to cause immediate device failures (e.g. components melting). Nonexhaustive fault injection testing has been used to evaluate the performance of the design. Note that practical considerations towards WSN implementation, such as energy provisioning, power consumption and communication protocols, are only partially considered as they are beyond the scope of this work.

1.4.3 Methods

This work is divided into four steps: first, redundancy management approaches have been investigated to identify potential techniques that can be used to improve fault coverage and to reduce the impact of CCFs. Next, a redundancy management system is designed. In the third step, this design is implemented in a WSN system. In the final step, the performance of the implemented design has been evaluated through reliability analysis, fault injection testing and an industrial test deployment.

1.5 Organization

The remainder of this work is organized as follows. Relevant literature on fault-tolerant systems, the impact of harsh environments on electronic components and redundancy management systems have been reviewed in Chapter 2. Modelling and analysis of imperfect fault coverage and CCFs have been discussed in Chapter 3. In Chapter 4, the redundancy management system design is presented. Chapter 5 has described the implementation of a prototype WSN system, and Chapter 6 has presented the evaluation results. Finally, the work has been concluded with a summary and the contributions in Chapter 7.

Chapter 2

2 Background & Literature Review

In this chapter, some background information on fault-tolerant systems and the impact of harsh environments on electronic components are discussed. Different approaches for a redundancy management system using BITs as well as existing WSNs for harsh environments are also reviewed.

2.1 Fault-Tolerant Systems

Fault-tolerance is a term used to describe the ability for a system to operate correctly, despite the presence of errors or faults [13]. Three core principles govern the various approaches for implementing fault-tolerance that can improve system reliability. A variety of methods exist to model and analyze a system's reliability. These methods and models, along with some literature on existing redundancy management systems, are discussed next.

2.1.1 Principles and Evaluation Metrics

Fault-tolerance is built upon three core design principles: redundancy, diversity and independence [13]. Each of these design principles can be used in conjunction with each other to enhance a system's reliability.

As mentioned previously, redundancy is the act of replicating critical components in a system, such that some components can remain as backups and assume operation only if the primary component fails. Redundancy is usually implemented for more critical components that either have a higher chance of failure or are essential for correct system operation. Redundancy can be active or passive [13]. Active redundancy describes when the redundant components operate concurrently, enabling immediate substitution of a faulty component. In passive redundancy, backup components remain in a standby state until needed.

Diversity in a design holds many similarities with redundancy as it relates to redundant components. A diverse component is one that is functionality equivalent such that it can

assume operation if the primary component fails, however its functionality is derived from different underlying mechanisms or construction. For example, powering an electronic device with a primary AC power supply and backup DC power supply would be considered as diverse. The advantage with using diverse backups occurs when failure modes are different between the components. Continuing with the power supply example, if an AC supply requires a power-grid connection, whereas a DC supply is powered through an external battery pack, then these two components have different failure modes. If the AC supply fails due to the loss of the grid connection, the DC supply would not be inherently impacted by this failure mode. To contrast this scenario, if the two power supplies depend upon a common set of voltage regulators that then becomes damaged, both power supplies can be impacted and fail simultaneously (note that failure here is defined as the inability to perform the intended operation). This type of failure scenario is often referred to as a CCF or a single point of failure.

Lastly, independence in design refers to the exclusion or separation of components such that a failure in one component does not impact the operation of the other components. A transformer is a common example of independence in design since certain types of faults are not directly translated between the primary and secondary windings.

As mentioned, reliability can be used to express the ability of a system to tolerate faults [12]. For a system with a constant failure rate, λ , the reliability function is described as

$$
p(t) = e^{-\lambda t},\tag{2.1}
$$

with t being the time and $p(t)$ being the reliability at a given time. The higher the reliability, the greater the chance that a system will be operational at time t . Another metric often associated with a system's reliability is mean time to failure (MTTF) [18] which is the expected time to failure. For a system with a constant failure rate, MTTF is defined as

$$
MTTF = \frac{1}{\lambda}.\tag{2.2}
$$

When comparing systems of varying complexity, a normalized MTTF timescale can be used, which is defined as

$$
normalized \, MTTF = \frac{t}{MTTF}.\tag{2.3}
$$

A normalized timescale can be used to compare reliability improvements in a design while abstracting away details that relate a system's specific failure rate [17].

To evaluate a system's reliability, a model must first be developed that sufficiently represents the failure characteristics of a system. One of the simplest methods of modelling reliability is with reliability block diagrams (RBDs) [18]. By identifying the various failure modes of a system's components, component failure rates and the system's architecture, a model that represents the relationship between component failures and a complete system failure can be developed. For example, [Figure 2.1](#page-24-0) illustrates a RBD for a WSN gateway device. In this diagram, there are five key components: a radio unit, a processor unit, a memory unit, a power management unit, and a network interface unit. The first four components have no replication and are therefore represented as a single block. The network interface unit, however, has *n* replicated units and is therefore represented as *n* blocks in parallel.

Figure 2.1: Example of a reliability block diagram for a WSN gateway device.

Another common method to evaluate a system's reliability is through the use of a fault tree [18]. A fault tree is similar to RBDs, but it is instead a more visual approach to represent a system based on its failure modes. To illustrate, a fault tree is shown in [Figure 2.2](#page-25-1) for a three-device WSN along with the WSN topology: a field device connected to a gateway device through a router. In [Figure 2.2,](#page-25-1) the system failure condition incorporates the WSN's

topology, as any device failure would result in a loss of information from the field device. Each device failure is composed of multiple initiating conditions as represented by the circles. An OR gate is used to relate the different failure conditions together, indicating that any failure condition results in a complete device failure. An AND gate is used to indicate that all initiating conditions must occur for that gate to be activated.

Figure 2.2: Fault tree for three WSN devices.

2.1.2 Impact of Harsh Environments on Electronic Components

As already stated in Chapter 1, harsh environments can negatively impact the reliability of a system by elevating the failure rates of components. Special consideration must be taken to accurately reflect an individual component's failure rate when developing a reliability model, since grossly inaccurate failure rates can render a model useless.

Most component manufacturers provide failure rate data for their various passive and active electronic components under an expected operating environment. Since industrial applications can have harsh and variable environmental conditions, this standard failure rate data alone may not be enough, and the failure rates may need to be estimated. For this

estimation, the manufacturer provided failure rates can be scaled by various degradation and acceleration factors under these alternate environmental conditions [19].

As mentioned, many factors contribute to harsh environment. In this work, reliability evaluation has been done only under elevated levels of temperatures and ionizing radiation that result in total ionizing dose (TID) effects. The most widespread approach to estimating the impact of elevated temperatures (below a level of a deterministic failure) on electronic components is the Arrhenius life-stress model [20],

$$
AF = e^{\left(\frac{E_{aa}}{k}\left(\frac{1}{T_1} - \frac{1}{T_2}\right)\right)}.
$$
\n
$$
(2.4)
$$

Here, AF is the acceleration factor, E_{aa} is the apparent activation energy, T_1 is the absolute temperature of test 1, T_2 is the absolute temperature of test 2, and k is Boltzmann's constant. This life-stress model can be used to scale the failure rate of an electronic component by the acceleration factor between two different test points. Manufacturers of electronic components usually provide a failure rate for their products at a given temperature, such as at 55° C. Given this information and the apparent activation energy, the acceleration factor for the component can be determined at a new temperature, enabling adjusted failure rate estimates.

Overall, the benefit of the Arrhenius model is two-fold: the first is that the anticipated failure rate of a component, and thus the reliability of a system, can be estimated at various ambient temperature levels if a full range of manufacturer provided data is unavailable. The second is that accelerated tests can be completed on a part at a high temperature for a short duration, and then extrapolated to estimate the failure rate at much lower temperatures.

More advanced multi-parameter Arrhenius models have been suggested for use that take into account multiple failure mechanisms for each component under study [20]. These models can result in more accurate estimates by using different activation energies for each failure mechanism. In practice, the use of these multi-parameter models can be challenging as manufacturers rarely provide a breakdown of a component's failure mechanisms.

MIL-HBDK-217 [21] is a military handbook produced to aid in determining accurate failure rate data for electronic components. Based on experimental and field data, this handbook provides scaling factors (including temperature induced acceleration factors) that can be used to adjust a component's failure rate to a wide variety of environmental conditions. MIL-HBDK-217 also provides scaling factor adjustments for military environments (such as on naval ships or when airborne). Based on the scaling method suggest in this handbook, failure rates are scaled as follows,

$$
\lambda' = \lambda \theta, \tag{2.5}
$$

where θ is the scaling term and λ' is the new component failure rate. Equations provided in MIL-HBDK-217 can be used to solve for θ based on the environmental conditions and the type of electronic component. Note that if the failure rate for a non-military environment is to be estimated using MIL-HBDK-217, the only scaling term would come from the temperature acceleration factor, resulting in

$$
\lambda' = \lambda AF. \tag{2.6}
$$

Neither MIL-HBDK-217 nor the more modern JEP122 account for the impact of ionizing radiation in the failure rates of electronic components. Instead, a second scaling term called the radiation degradation factor, Δ_k , is required to estimate the negative impact of this harsh environmental condition [22]. With the use of Δ_k , the failure probability of an electronic component can be scaled to estimate the new failure probability after receiving a specified TID,

$$
p(t)' = (1 - \Delta_k)e^{-\lambda t}.\tag{2.7}
$$

Similar to E_{aa} , the radiation degradation factor is experimentally determined. The work in [23] along with the NASA Goddard Space Flight Center radiation test database [24] can be used to estimate this degradation factor, thus enabling reliability estimates under varying levels of ionizing radiation.

Both scaling techniques can be combined as follows to provide a more accurate reliability approximation under an environment with higher levels of temperature and radiation,

$$
p(t)' = (1 - \Delta_k)e^{-\lambda' t}, \qquad (2.8)
$$

with $p(t)$ being the approximated component reliability.

2.1.3 Redundancy Management Systems Using the BIT Approach

The primary issue with the BIT approach is imperfect fault coverage [14]. Coverage describes the probability that a fault will be correctly identified through some protection scheme. Perfect fault coverage results in most voting-based systems where a comparison between multiple outputs can detect the occurrence of a fault. On the other hand, BITs have imperfect fault coverage due to the difficulties that arise when detecting a fault through some sort of self-test. Even with high levels of coverage (nearing 99%), a significant negative impact on a system's reliability occurs [14]. Therefore, circumventing this imperfect coverage issue is the primary challenge faced with BITs.

BIT coverage can be improved by enhancing fault detection capabilities. A significant portion of the literature has focused on fault detection and identification by using a variety of techniques that fall under model-based, signal-based, knowledge-based, and hybrid approaches [25] [26]. Although these approaches are powerful in the correct setting, their use is relatively limited for detecting embedded systems faults in a WSN device.

A BIT-based system structure that uses a comparator has been proposed in [27]. In their work, comparison logic (similar to voting logic) and BITs are used simultaneously in a dual-redundant system. [Figure 2.3](#page-29-2) illustrates this arrangement. The advantage of the comparison logic is that a fail-safe mechanism is introduced that can halt a system's operations under certain output conditions. This fail-safe mechanism, however, does not improve a system's fault coverage, since the system ceases to operate when this mechanism is activated.

Figure 2.3: Redundancy management with BITs and comparison logic.

The other major challenge for the BIT approach is the introduction of additional CCF mechanisms. If, for example, a switch that is used to select a redundant system's output becomes damaged, then this faulty switch could result in a complete system failure. This added risk introduced by the redundancy management system in some cases can reduce a system's reliability rather than improve it [17]. To better understand the relationship between a potential reliability improvement from a redundant design, a redundancyrelevance boundary has been proposed in [17]. Their work emphasizes the importance of considering how CCFs can impact a redundant design and has been discussed further in Chapter 3.

2.2 WSNs for Harsh Environments

There are a variety of target applications for WSN devices in harsh environments, from industrial use to disaster relief scenarios. In this section, WSN device-level architectures and existing techniques employed in WSNs for harsh environment applications are reviewed.

2.2.1 Device-Level Architecture

WSN devices consist of several key components, as shown in [Figure 2.4.](#page-30-0) Field devices consist of a radio unit, processor unit, memory unit, power management unit, and a sensor interface unit [28] [29]. Routing devices are similar to the field devices, except that the sensor interface unit may be excluded. Gateway devices, also referred to as sink nodes, are the end destination for data collected by WSN devices. Since gateways accommodate a high volume of traffic and must interface externally to a backbone network, these devices are usually equipped with additional resources such as Wi-Fi chips, Ethernet ports and local server capabilities. Gateways usually require some infrastructure such as line power, Internet access or cellular connectivity to support these resources. Typical components of a gateway device are shown in [Figure 2.5.](#page-30-1)

Figure 2.4: Typical components of a WSN field device.

Figure 2.5: Typical components of a WSN gateway device.

2.2.2 Existing Monitoring Systems in Harsh Environments

Some work has been done to develop different WSN systems that are more suited for use in disaster response scenarios, as summarized in [30]. Seven different system topologies have been identified that make use of multiple wireless communication technologies such as satellite, wide area networks and personal area networks. The use of multiple wireless technologies and wireless channels within a single system can improve information availability in a network [22]. Some other work has proposed the use of multi-radio WSN devices [31] to allow for the integration of multiple WSN technologies on one board (such as Bluetooth and Zigbee).

There are some custom-made, robust wireless monitoring systems developed primarily for military applications [32], [33]. These systems include casings or shielding to provide protection against adverse environmental conditions. The primary objective for these monitoring systems is reliable, long-range communication. Therefore, these systems transmit information directly to their end devices by using satellite, cellular or other longrange technologies.

Other wireless monitoring solutions rely upon the use of advanced materials and simple circuits that are less susceptible to failures in harsh environments. In [34], a specialized wireless telemetry system has been developed that can withstand temperatures greater than 350°C. Similarly, a wireless pressure sensing solution for high temperatures has been developed in [35] that relies upon a simple circuit design and FM technology.

A recent work has proposed a wireless monitoring system for NPPs under severe accident conditions [23]. Their objective has been to design a radiation-tolerant system using commercial off-the-shelf components. The system has used radiation shielding and a redundant design that is based on voting logic.

2.3 Limitations of Existing Work

In general, existing WSN systems for harsh environments use a combination of protective casing for the devices and system level fault-tolerance (such as ensuring redundant communication paths in the network). Protective casings create a physical barrier between devices and the harsh environment, reducing the environment's negative impact on a device's reliability. However, their effectiveness can be limited. For example, elevated ambient temperatures and certain types of ionizing radiation can penetrate through protective casing. Although, in general, protective casing can reduce the impact of harsh environment to a certain degree, the penetrating effect may still result in conditions that are higher than the typical operating environment for a WSN device. Applying additional protection to a system to mitigate this effect may not be practical. For example, radiation shielding can be used to reduce the TID received by a WSN device, but radiation shielding can be an expensive and heavy solution.

The topologies summarized in [22] can enhance system level fault-tolerance by incorporating multiple wireless communication technologies into a WSN. For these topologies to be effective, the deployment of their nodes must be restricted to either a strategic or dense deployment. This restricted deployment may not be achievable in certain applications, limiting their applicability.

Many of the wireless monitoring systems developed for military applications use longrange communication technologies that can have difficulty in indoor, industrial environments. For example, wireless signals may not be able to penetrate through the thick concrete walls of a containment building in a NPP.

The specialized wireless monitoring systems with advanced materials for use in high temperature applications rely upon simple RF circuit technology for point-to-point communication. These systems may not be suitable for use in harsh environments if the sensor information cannot be transmitted directly to a base station. For example, within the containment building in a NPP, mesh networking might be the only option to relay environmental data wirelessly to a sink device. These types of systems are also not developed for environments with high levels of ionizing radiation.

In [23], device level fault-tolerance has been achieved with a redundant design based on the voting logic approach. Their design has focused on radiation-tolerant design in the circuit and system level.

Overall, protective casings and system level fault-tolerance techniques can have a limited effectiveness in certain applications and deployments. Device level fault-tolerance by incorporating redundancy in a design can be used to improve WSN system reliability. Device level fault-tolerance can also be used in addition to protective casing and system level fault-tolerance to further improve WSN reliability for certain critical applications, such as monitoring an industrial plant during an accident condition. For device level faulttolerance from a redundant design to be an effective solution, the issues of imperfect fault coverage and CCFs have to be addressed. The remainder of this work details the development of a BIT-based redundancy management system that address these two issues.

Chapter 3

3 Modelling Imperfect Fault Coverage

The design of a redundancy management system for WSN devices begins with the development of an appropriate reliability model. In this chapter, imperfect fault coverage is first modelled under ideal conditions, in which no CCFs are introduced. An approach to improve coverage through the use of a *supervisory unit* is then presented. Afterwards, a more advanced reliability model is developed that includes the impact of both imperfect fault coverage and CCFs. Finally, a modularized architecture that could further diminish some of the negative aspects of CCFs is discussed.

3.1 Modelling Imperfect Fault Coverage Without CCFs

In the development of a model for a redundant system, the following assumptions are:

- The failure rate λ is constant in a given operating environment.
- Redundant components are in an active state.
- Redundant components are identical, such that $\lambda_1 = \lambda_2 = \lambda_n = \lambda$.
- Failures are independent and permanent among all components.

Note that dependant component failures are modelled separately as CCFs in Section 3.2. As well, component failure rates could be time-dependant if environmental conditions (such as temperature) change. Therefore, the impact of a non-constant failure rates on the developed models will be discussed throughout this chapter when relevant.

Under the previous assumptions, the reliability for a device consisting of a single component can be derived from an exponential distribution as detailed in [14],

$$
p(t) = e^{-\lambda t} \tag{3.1}
$$

If a single device requires two identical components to be functioning at the same time, it is not redundant. The corresponding reliability model would simply be the product of reliability of the two components,

$$
R(t) = e^{-\lambda t}e^{-\lambda t} = e^{-2\lambda t}.
$$
\n(3.2)

Instead, if a single device consists of two identical components, and only one component needed to be operational for the device to work, then the device would be described as dual-redundant. The reliability model for such a device relates to the parallel product of the reliability of each component [14],

$$
R(t) = 1 - (1 - e^{-\lambda t})(1 - e^{-\lambda t}) = 1 - (1 - e^{-\lambda t})^2.
$$
 (3.3)

In general, the reliability for an n -redundant device is

$$
R(n,t) = 1 - \left(1 - e^{-\lambda t}\right)^n.
$$
 (3.4)

Equation [\(3.4\)](#page-35-0) assumes perfect fault coverage (or that a fault does not need to be detected for the system to continue to operate). As noted in Section 2.1.3, this perfect level of coverage is usually not achievable in a BIT based approach.

The BIT approach operates on a 1-out-of- n basis [14], meaning that the system can continue to work if at least 1 replicated component is still functional. The reliability for such a system is given as

$$
R(n, p(t)) = \sum_{i=1}^{n} {n \choose i} p^{i} (1-p)^{n-i}
$$
\n(3.5)

where R is the device reliability, and p is the component reliability. In contrast, the general model for a voting logic system operating on a 2-out-of- n basis is

$$
R(n, p(t)) = \sum_{i=2}^{n} {n \choose 1} p^{i} (1-p)^{n-i}.
$$
 (3.6)

One method to express imperfect fault coverage is to separate a component's failure rate into covered and uncovered faults as such:

$$
\lambda = \lambda_c + \lambda_{uc},\tag{3.7}
$$

where λ_c is the covered fault failure rate and λ_{uc} is the uncovered fault failure rate [14]. The entire system's failure rate, λ , can be split into the faults that can be detected, λ_c , and the faults that cannot be detected, λ_{uc} . Coincidentally, in this work, a covered fault can be detected whereas an uncovered fault cannot be detected. An alternate approach to express fault coverage is through a component's coverage ratio, c [14]. Following from Equation [\(3.7\),](#page-36-0) the relationship between the covered and uncovered failure rate is

$$
\lambda_c = c\lambda,\tag{3.8}
$$

and

$$
\lambda_{uc} = (1 - c)\lambda. \tag{3.9}
$$

Before developing the reliability model for a system with imperfect fault coverage, first the reliability impact of λ_c and λ_{uc} needs to be understood. If, for example, a detectable fault (λ_c) has occurred in one component of a dual-redundant device, that device could substitute the correctly operating component for the faulty component. If, instead, an undetectable fault (λ_{uc}) has occurred in the same device, that fault would go unmitigated in the system and the device would enter into a failed state.

Fault coverage can be incorporated into the reliability model developed in Equation [\(3.4\),](#page-35-0) as detailed in [14], yielding

$$
R(n, p(t), c) = \sum_{i=1}^{n} cT(i, c) \binom{n}{1} p^{i} (1-p)^{n-i}, \qquad (3.10)
$$

where $cT(i, c)$ is the set of products of the k-subset of the coverage ratio vector, c , with exactly $n - i$ elements. The coverage ratio vector for an *n*-redundant system is $c =$ $\{c_1, ..., c_n\}$. For a triple-redundant system ($n = 3$), the cT set would be

$$
cT(1, c) = \{c_1c_2, c_1c_3, c_2c_3\}
$$

\n
$$
cT(2, c) = \{c_1, c_2, c_3\}
$$

\n
$$
cT(3, c) = \{1\}.
$$

Each component has been assumed to be identical,

$$
c_1 = c_2 = c_n = c. \t\t(3.11)
$$

From Equation [\(3.10\)](#page-36-1) it can be seen that the uncovered faults negatively impact the reliability of the system, with a smaller fault coverage leading to a reduction in reliability. If $c = 0$ in a dual-redundant system, the model reverts back to Equation [\(3.2\)](#page-35-1) where a failure in either component results in a device failure. If $c = 1$, the system reverts to Equation [\(3.3\),](#page-35-2) a dual-redundant system with perfect coverage.

To illustrate this, [Figure 3.1](#page-37-0) depicts the reliability curve for a dual-redundant system under imperfect fault coverage conditions. The y-axis represents the reliability, *, for a dual*redundant device, and the x-axis represents the time t normalized by the MTTF for a single component system. Perfect coverage is when $c = 1$, and imperfect coverage is when $0 <$ $c < 1$. A clear observation from [Figure 3.1](#page-37-0) is that as the coverage ratio decreases, the reliability diminishes. Conversely, improving the coverage ratio would improve the system's reliability.

Figure 3.1: System reliability under imperfect fault coverage.

The results in [Figure 3.1](#page-37-0) can be explored further to better understand the significance of imperfect fault coverage on a system. A question that can be raised is whether the use of redundancy in a system can be harmful rather than beneficial? To answer this question, a redundancy-relevance boundary for a BIT-based system with a varying level of redundancy (LR) and imperfect fault coverage has been developed. This boundary is shown in [Figure](#page-38-0) [3.2.](#page-38-0) Here, a redundancy level of 0 represents a system with no redundancy, whereas a redundancy level of 1 represents a dual-redundant system.

Figure 3.2: Redundancy-relevance boundary for a BIT-based system.

This boundary shows the minimum coverage level required for a redundant system to improve the MTTF relative to a non-redundant system. For a dual-redundant system (with LR=1), a coverage ratio greater than 0.5 is required. To contrast, a triple-redundant system requires the coverage ratio to be greater than 0.605. Ensuring that the coverage ratio is larger than this boundary condition is imperative to successfully improve the reliability in a redundant system.

Recall the discussion of the comparator block presented in Section 2.1.3. It has been identified that the weakness of this supplementary detection system is its inability to improve fault coverage. What if instead this comparator block served a dual-purpose, capable of detecting a malfunction and providing additional means to identify the fault? If this supplementary block is more capable (perhaps able to provide additional information to the existing BITs or providing new identification mechanisms), it could improve a system's fault coverage, which in turn, improves the reliability.

With this idea in mind, the notion of a *supervisory unit* with the ability to detect such malfunctions is proposed. This supervisory unit provides the functionality of a comparator block while also contributing to an improved fault coverage. This proposed topology, shown in [Figure 3.3,](#page-39-0) still relies upon the BIT approach as its redundancy management scheme. The supervisory unit improves fault coverage by providing feedback to the existing BIT mechanisms upon the detection of a malfunction. The first step to fault identification is, after all, detecting an issue. This information can then help to trigger additional tests within each of the replicated component's BITs to help with fault identification.

Figure 3.3: Modified BIT topology to include a supervisory unit.

To see if such a topology can, in theory, improve reliability, consider the following example. Imagine a dual-redundant system where each replicated component computes a number based on its input that then sends this value as a signal through a communication channel. Before this output value is sent, a BIT re-checks that the computed value is correct. Unknown to the BIT, however, is that noise might occasionally interfere with only one of the output signals being sent from the two components, which coincidentally changes the output value. If an intermediate element (the proposed supervisory unit) has received this signal, a mismatch between the two components could be detected when noise interferes with the signal. When this situation occurs, this fault information is fed back to each component's BIT. By feeding back the corresponding output signals, a BIT could then conclude that the signal received is not the signal intended to be sent. In turn, the component with the noisy communication line could be identified and isolated, allowing the alternate component to resume operation.

Quantitively, a supervisory unit can be incorporated into a reliability model to study its impact. A new variable denotes the added fault coverage provided by the supervisory unit, called the supervisory coverage ratio, c_s . This supervisory coverage ratio provides an additive effect with the existing system's original coverage, c . For example, if a system has a coverage ratio of $c = 0.5$ and the supervisory unit can detect and identify an additional 10% of faults, then $c_s = 0.1$. The system's new coverage ratio, c', would then be

$$
c' = c + c_s. \tag{3.12}
$$

A new reliability model can be produced that includes this additive coverage effect,

$$
R(n, p(t), c') = \sum_{i=1}^{n} cT(i, c') \left(\frac{n}{1}\right) p^{i} (1-p)^{n-i}.
$$
 (3.13)

This model, however, is not yet complete since the supervisory unit (like all components) can fail. As well, the supervisory unit can also provide false information. It would therefore have a corresponding failure rate, λ_s , and a false positive rate, λ_f , along with its own MTTF, denoted by MTTFs. The replicated components have their MTTF denoted by MTTF_c. If the supervisory unit is assumed to be fail-safe (it cannot result in a CCF and any false positives can be corrected for, such that $\lambda_f = \lambda_s$, its failure cannot result in a system failure, but the added coverage improvement would be lost.

How can this loss of coverage condition that arises when the supervisory unit suffers from a failure be effectively incorporate into a reliability model? In a sense, such behaviour dynamically alters a reliability model that can complicate matters.

A simple approach can be taken to estimate its effects. Consider the following three situations:

- 1. The supervisory unit cannot fail (or is highly unlikely to fail).
- 2. The supervisory unit has a similar failure rate to the replicated components.
- 3. The supervisory unit will fail significantly sooner than the replicated components.

In the first situation, it would be expected that the full fault coverage benefit of the system can be obtained. That is, the reliability model would be Equation [\(3.13\).](#page-40-0)

In the second situation, at some point the supervisory unit will fail. It would therefore make sense that initially a reliability improvement is achieved close to the maximum achievable improvement from Equation [\(3.13\).](#page-40-0) As time progresses, however, the likelihood that the supervisory unit has failed increases. The reliability should therefore rest somewhere below the upper bound. As time extends out further, the system would perform as if no supervisory unit has been added and approach the lower bound in Equation [\(3.10\).](#page-36-1)

In the third situation, it would be expected that the supervisory unit provides a marginal improvement since its failure should occur rather quickly. The system's reliability should be expected to quickly approach the lower bound in Equation [\(3.10\).](#page-36-1)

From these three scenarios, it can be concluded that the proposed system's reliability should be bounded at all times by Equation [\(3.10\)](#page-36-1) and Equation [\(3.13\).](#page-40-0) Further, the reliability improvement depends upon the supervisory unit's failure rate, λ_s and the replicated component's, λ . A decay in the reliability improvement is expected as time progresses based on some ratio of these two failure rates.

A reliability model has been developed that satisfies the previous conditions:

$$
R(n, p(t), c'(a)) = \sum_{i=1}^{n} cT(i, c')) \left(\frac{n}{1}\right) p^{i} (1-p)^{n-i}, \qquad (3.14)
$$

where α is the coverage decay function

$$
\alpha = e^{-\frac{MTTF_c}{MTTF_s}} = e^{-\gamma} \tag{3.15}
$$

and c' is

$$
c'(\alpha) = c + \alpha c_s. \tag{3.16}
$$

Note that γ is the ratio of the two MTTFs. The effect of this decay function on the reliability model is shown in [Figure 3.4.](#page-42-0) Note that the model's reliability satisfies the previously described conditions, as it is effectively bounded between Equation [\(3.10\)](#page-36-1) and Equation $(3.13).$

Figure 3.4: Bounding effect of the coverage decay function.

Note here that the model developed in Equation [\(3.14\)](#page-41-0) is merely an estimate to better understand the impact of a fail-safe supervisory unit that is subject to failures. As such, a rigours proof is not provided or needed to gather insight towards the value of the supervisory unit in a redundant design. Ultimately, for this model to be deemed correct (either as conservative or optimistic) additional testing is required. Nevertheless, insight towards the desirable traits of the postulated supervisory unit can be extracted.

The first insight from [Figure 3.4](#page-42-0) is that if the supervisory unit is fail-safe, then no harm can come to the system's overall reliability. Second is that the failure rate of the supervisory unit influences its reliability improvement. Ideally, a failure rate much smaller than the component's rate would yield the greatest improvement. If these two failure rates are similar to each other, a smaller yet significant reliability improvement is gained.

Note that in scenarios with non-constant failure rates (such as when an environment changes), it is important to ensure that the supervisory unit's failure rate is equal to or lower than the component's failure rate. To determine whether this condition is met, failure rates can be calculated under a variety of expected environmental conditions using the scaling factors presented in Section 2.1.2.

In summary, an alternate BIT topology has been proposed that uses a supervisory unit to improve fault coverage, and ergo, reliability. This reliability improvement hinges on the failure rate of the added unit. Both the BITs and the supervisory unit can introduce additional failure mechanisms in a system that can negatively impact the reliability. These considerations are discussed next.

3.2 Modelling Imperfect Fault Coverage with CCFs

The previous model has not considered the impact of any additional CCFs. A more realistic reliability model needs to be developed that includes this fact. The objective of such a model is to identify the boundary condition in which a system's reliability is improved given the model parameters.

One conservative approach for modelling CCFs in a redundant design is the β -factor model [36]. The β -factor is a single parameter approach to modelling the probability of an event occurring, and is defined as the ratio of the CCF rate to the total failure rate of the components,

$$
\beta = \frac{\lambda_{CFF}}{\lambda + \lambda_{CCF}}.
$$
\n(3.17)

This β -factor can be included in the previously derived reliability model for a redundant system,

$$
R(n, p(t), \beta, \lambda) = \left(\sum_{i=1}^{n} {n \choose i} p^{i} (1-p)^{n-i}\right) e^{-\frac{\beta}{1-\beta}\lambda t}.
$$
 (3.18)

Equation [\(3.18\)](#page-44-0) represents the reliability model for an idealized redundant system. For a system constructed with the BIT approach, the respective reliability model is

$$
R(n, p(t), \mathbf{c}, \beta, \lambda) = \left(\sum_{i=1}^{n} cT(i, c)\left(\frac{n}{i}\right)p^{i}(1-p)^{n-i}\right)e^{-\frac{\beta}{1-\beta}\lambda t}, \tag{3.19}
$$

whereas the voting logic approach would have

$$
R(n, p(t), \beta, \lambda) = \left(\sum_{i=2}^{n} \binom{n}{i} p^i (1-p)^{n-i}\right) e^{-\frac{\beta}{1-\beta}\lambda t}
$$
(3.20)

as its reliability model.

A similar redundancy-relevance boundary to that in [Figure 3.2](#page-38-0) can be developed to compare the effective reliability improvement of these two approaches. This boundary is shown in [Figure 3.5](#page-45-0) under a varying level of redundancy.

Figure 3.5: Redundancy-relevance boundary for both redundancy approaches.

As shown in [Figure 3.5,](#page-45-0) there is a maximum value of the β -factor for each level of redundancy after which the reliability, in terms of the MTTF, would decrease rather than improve. For a BIT-based triple-redundant system, this value is 0.379. In contrast, for a triple-redundant voting logic-based approach, even with $\beta = 0$ (no CCFs), a MTTF-based reliability improvement cannot be achieved. This result illustrates as to why voting logic might not be as well suited for certain non-repairable monitoring applications using WSN systems. For example, if a voting logic system is non-repairable and is intended to operate until failure, it is expected that this system would fail sooner than a non-redundant system. The cause of this earlier failure stems from the increased number of components in a redundant system that increases the occurrence of component failures within the same time interval. Coincidentally, redundant components decrease the mean time between failure (MTBF) within a system. Using a triple-redundant system operating on a 2-out-of- n basis

as an example, this decreased MTBF means that two components are expected to fail within the normalized MTTF interval, reducing this system's MTTF.

A reliability model can now be produced that incorporates both imperfect fault coverage and CCFs for BIT-based systems as follows,

$$
R(n, p(t), c, \beta, \lambda) = \left(\sum_{i=1}^n cT(i, c)\left(\frac{n}{1}\right)p^i(1-p)^{n-i}\right)e^{-\frac{\beta}{1-\beta}\lambda t}.
$$
 (3.21)

Based on this model, a more advanced redundancy-relevance boundary can be developed. This new boundary is shown in [Figure 3.6.](#page-46-0) Note that the reliability reduction that results from imperfect fault coverage and CCFs is additive; a reduction in fault coverage necessitates an improvement in the β -factor, and vice versa.

Figure 3.6: Advanced redundancy-relevance boundary considering imperfect fault coverage and CCFs.

This boundary can be used as a design tool, enabling quick reliability analysis of a redundant system. Given the estimates for a system's fault coverage, its β -factor, and the level of redundancy, a designer can determine whether a reliability improvement is achieved.

The redundancy-relevance boundary in its current form does not indicate the magnitude of the reliability improvement. Once a design is deemed to improve reliability, it is desirable to determine the level of improvement.

In this regard, a *reliability-improvement plane* for each level of redundancy can be produced from Equation [\(3.21\).](#page-46-1) By normalizing the factor of improvement against a nonredundant system's MTTF, the relative improvement can be determined. [Figure 3.7](#page-48-0) shows this relative improvement under different levels of redundancy.

These individual planes can be used to determine the anticipated reliability improvement given the appropriate model parameters. For example, a triple-redundant system's level of redundancy is two. By examining the top right plane in [Figure 3.7,](#page-48-0) it can be seen that the maximum MTTF improvement is 1.83 times greater than that of a non-redundant system. This improvement is only attained when the coverage is equal to one and the β -factor is equal to zero.

The proposed supervisory unit now can be integrated with the reliability model produced in Equation [\(3.21\).](#page-46-1) Previously, the supervisory unit has been assumed to be fail-safe; that assumption can now be removed. Note, however, that false positives are still assumed to be corrected for.

Figure 3.7: Reliability-improvement planes for different levels of redundancy.

First, the supervisor β -factor, β_s , is defined as

$$
\beta_s = e^{-\frac{\beta_s}{1-\beta_s}\lambda_s t}.\tag{3.22}
$$

Also, to help simplify the final model, the failure rate of the supervisor unit can be defined in terms of the replicated component's failure rate,

$$
\frac{\lambda_s}{\lambda} = \gamma,\tag{3.23}
$$

with γ being the ratio of the failure rates. Combining Equation [\(3.22\)](#page-48-1) and Equation [\(3.23\)](#page-48-2) yields

$$
\beta_s = e^{-\frac{\beta_s}{1-\beta_s}\gamma \lambda t}.
$$
\n(3.24)

This β_s -factor, along with the fault coverage improvement, can be integrated into Equation [\(3.21\)](#page-46-1) as,

$$
R(n, p(t), c'(\alpha), \beta, \lambda, \beta_s, \gamma) = \left(\sum_{i=1}^n cT(i, c')\right) \left(\frac{n}{1}\right) p^i (1 - \gamma)^{n-i}
$$
\n
$$
p)^{n-i} e^{-\frac{\beta}{1-\beta}\lambda t} e^{-\frac{\beta_s}{1-\beta_s}\gamma \lambda t},
$$
\n(3.25)

which simplifies to

$$
R(n, p(t), c'(\alpha), \beta, \lambda, \beta_s, \gamma) = \left(\sum_{i=1}^n cT(i, c')\right) \left(\frac{n}{1}\right) p^i (1 - p)^{n-i}
$$
\n
$$
(3.26)
$$
\n
$$
p)^{n-i} e^{-\lambda t \left(\frac{\beta}{1-\beta} + \frac{\gamma \beta_s}{1-\beta_s}\right)}.
$$

Equation [\(3.26\)](#page-49-0) helps identify the final design considerations to determine whether the inclusion of this supervisory unit is indeed beneficial.

Both the benefit received from the supervisory coverage ratio, c_s , and the consequence of the β_s -factor are dependent upon γ . This dependency indicates that if the failure rate of the supervisory unit is sufficiently smaller than the replicated component's failure rate, the inclusion of this unit can be justified. The following case study further elaborates this effect.

[Figure 3.8](#page-50-0) illustrates the reliability-improvement plane for a dual-redundant system with a varying γ and β_s -factor. In this scenario, $c = 0.7$, $c_s = 0$, and $\beta = 0$ (the effects of c_s and β have been removed for clarity). Note that when γ is small, so is the negative impact from β_s . This result indicates that if the failure rate of the supervisory unit is significantly smaller than that of the replicated component's rate, its potential to harm the redundant system is reduced. In contrast, as λ_s grows larger than λ , a modest β_s -factor can drastically alter a system's reliability.

Figure 3.8: Reliability-improvement plane for a supervisory unit system under a varying γ and β_s -factor.

From the consequence illustrated in [Figure 3.8,](#page-50-0) the final requirement for the supervisory unit can be identified. That is, the failure rate of the supervisor, λ_s , should be smaller than the failure rate of the replicated components, λ . Note that if the failure rates are not constant, then the supervisory failure rate should be smaller than the replicated component's failure rate during the entire mission time.

Altogether, several design requirements have been identified from the previously developed models to determine whether the introduction of a supervisory unit can benefit a redundant design:

- A small γ ($\lambda_s < \lambda$) results in a larger fault coverage improvement from c_s .
- If the supervisory unit is fail-safe, its use can only improve reliability.
- If the supervisory unit is not fail-safe, a smaller γ allows for a larger β_s -factor.

Of course, a reliability improvement would only be plausible if the requirements from the redundancy-relevance boundaries are also satisfied.

3.3 Impact of Modularity on CCFs

So far, the primary benefit of the proposed supervisor unit has been seen to be a potential increase for fault coverage. Issues with CCFs have yet to be addressed. It would be beneficial if the negative impact from CCFs could be reduced.

Suppose that the entire model for a dual-redundant system with a supervisory unit is treated as an individual system module and then replicated, as shown in [Figure 3.9.](#page-51-0) What are the benefits and drawbacks of such an approach?

The leading benefit of the modularized, dual-redundant system architecture can potentially be the reliability improvement from this second layer of redundancy. The two main drawbacks are an increased number of resources used (4 components and 2 supervisory units) and risk of CCFs. These aspects are explored next.

Figure 3.9: Modularized dual-redundant system topology.

First, the reliability improvement from the modularized system against resources used is analyzed. A quadruple-redundant voting system can be produced, as shown in [Figure 3.10,](#page-52-0) using the same number resources as in the BIT system. Note that the voting system does not suffer from the imperfect fault coverage problem since these types of redundant systems can have perfect or near-perfect fault coverage [14].

Figure 3.10: Quadruple-redundant system using voting logic.

However, as mentioned earlier in Section 3.1, there is a downside to the voting logic approach. The 2-out-of- n requirement can produce long-term reliability issues due to its

[Figure](#page-54-0) 3.11 illustrates this by comparing system reliability under varying degrees of redundancy. Shown is the reliability for multiple idealized systems, against the reliability for a 2-out-of-4 voting logic system. Note that the ideal system is one where the redundancy management approach has perfect fault detection and cannot fail (see Equation [\(3.4\)\)](#page-35-0).

Figure 3.11: Comparison of reliability for different topologies.

Although the voting logic system does provide a significant reliability improvement during the first half of its 'usable life', this improvement rapidly decays to levels lower than a nonredundant system. Due to this rapid decay, only an 8% improvement in the MTTF is gained. Voting logic operating on a 2-out-of- n basis is, therefore, not inherently suited for applications in non-repairable system with a long mission time (relative to the normalizing MTTF) if a low level of redundancy is used.

A reliability model can be produced that represents the proposed BIT-based modularized and dual-redundant system as follows,

$$
R_M = 1 - (1 - R)^2 e^{-\frac{\beta_M}{1 - \beta_M} \lambda t}, \qquad (3.27)
$$

where R_M is the redundancy of the modularized system and β_M corresponds to the CCFs by the reconfiguration mechanism shown in [Figure 3.9.](#page-51-0) For convenience, it is assumed that the modularized reconfiguration mechanisms are similar to that within each module, meaning that the β_M -factor could be approximated to

$$
\beta_M \cong \beta. \tag{3.28}
$$

Equation [\(3.27\)](#page-54-1) can then be re-written as

$$
R_M = 1 - (1 - R)^2 e^{-\frac{\beta}{1 - \beta}\lambda t}.
$$
 (3.29)

Usually, each time a problem is solved with redundancy, the risk of CCFs counteracts the reliability improvement. However, in this instance, the negative impact that results from the β -factor is indeed reduced.

To illustrate, [Figure 3.12](#page-56-0) shows two reliability-improvement planes for the proposed modularized system under a varying β -factor and coverage, c. For clarity, $c_s = 0$ and $\gamma =$ 1. The top plane is for when $\beta_s = 0.05$, whereas the bottom plane is for when $\beta_s = 0.15$.

From [Figure 3.12,](#page-56-0) it can be seen that a significant reliability improvement can be achieved in the modularized system over the non-modularized system across a wide range of model parameters. In both examples in [Figure 3.12,](#page-56-0) the modularized system can have an improved reliability with a coverage ratio of 0.6 and a β -factor of 0.1, whereas the non-modularized system cannot. Coincidentally, such a system can improve reliability even with a considerably low coverage ratio. The negative impact from the β_s -factor does, however, does still contribute to a reliability reduction.

For a comparison, [Figure 3.13](#page-57-0) shows the reliability-improvement plane for two systems under similar conditions but with $\gamma = 0.1$. This lower γ value means that the supervisory unit's failure rate is smaller by a factor of 10 than the replicated component's failure rate. It can be observed by comparing the two planes in [Figure 3.13](#page-57-0) that the negative impact from the β_s -factor has diminished.

Figure 3.12: Reliability-improvement plane for the dual-redundant system versus the modularized system with γ=1.

Figure 3.13: Reliability-improvement plane for the dual-redundant system versus the modularized system with γ=0.1.

The analysis of the modularized dual-redundant system shows that a significant reliability improvement can be gained under certain model parameters. That is, more variability for both β and c are allowed. When γ is small, more variability is also allowed for β_s . Thus, the modularized approach can help to reduce the negative impact from CCFs.

3.4 Impact of Diversity in Design on CCFs

The entire discussion so far has assumed the replicated components are identical, having the same failure mechanisms and failure rates. If the replicated components are still functionality equivalent but differ in their failure modes, it could be possible to reduce the risk of CCFs between the modularized system. Yet, it is still desirable that each of the diverse components are as reliable as each other so that no single component performs considerably worse.

Diversity in design is one common technique used to improve the reliability of a redundant system [18]. If non-overlapping failure modes exist between all of the redundant elements in a design, then it is possible to reduce the β -factor for a dual-redundant design, as well as the β_M -factor for a modularized system. Any reduction in either of these parameters would increase reliability. Therefore, implementing diversity in design is a second strategy to reduce the impact of CCFs in a redundant design.

3.5 Summary of Considerations

The impact of imperfect fault coverage and CCFs have been explored on several different redundant device topologies. It has been shown that if the proposed supervisory unit has a failure rate similar to or lower than that of the redundant components, it can increase fault coverage in a design. Further, it has been also shown that a modularized dual-redundant system architecture and diversity in design can be used to alleviate the reliability reduction that may be caused by CCFs. These considerations have guided the design for a redundancy management system, as discussed in Chapter 4.

Chapter 4

4 Redundancy Management System Design

With the foundation set for the conditions in which various topologies can improve a device's reliability, the proposed redundancy management system can now be developed. This redundancy management system consists of four parts: the device topology, a microcontroller-BIT, a supervisory diagnostics algorithm (SDA), and supplementary fault detection hardware (FDH). The combination of these four parts yields the complete redundancy management system design that has been implemented and evaluated in a WSN device.

4.1 Device Topology

The first part of the redundancy management system is the device topology. Based on the reliability models developed in Chapter 3, a diverse, modularized and dual-redundant topology with a supplementary supervisory unit has been selected. This topology is shown in [Figure 4.1.](#page-59-0)

Figure 4.1: Proposed redundancy management system topology.

Two diverse WSN modules have been selected, one at 900MHz (subsequently referred to as Module A), and one at 500MHz (subsequently referred to as Module B), each consisting of dual-redundant wireless microcontrollers (WMCUs) and a supervisory unit. Within a single module, the two selected WMCUs are diverse yet functionality equivalent (as described in Section 3.4). Additionally, each component within a module are interfaced together via a dual-redundant and diverse communication bus. Each module also uses a dual-redundant sensor interface for its input (noted in [Figure 4.1\)](#page-59-0). Note that from an electrical perspective, there is little difference in the presented system topology and the one analyzed in [Figure 3.9.](#page-51-0)

Each WMCU has its own BIT that is responsible for detecting faults. If a fault is detected, that component is deactivated. Also note that each WMCU has its own supplementary fault detection hardware. This hardware can help to detect digital bus communication faults, improving fault coverage for each component. Electromechanical relays act as switches within this fault detection hardware, isolating the faulty components electrically from other components. Finally, each supervisory unit has a supervisory diagnostic algorithm (SDA). The SDA is responsible for providing feedback to each of the replicated component's BIT, further assisting with faulty component identification.

4.2 Microcontroller-Based Built-in Test

The primary mechanisms for fault detection and identification is through software-based BITs that directly influence the fault coverage for each WMCU. Software BITs follow the traditional techniques to detect and identify faults within MCUs [37] [38] [39]. Examples of such techniques include watchdog timers, exception handlers and IO validation. The only unique aspect of these BITs is their ability to use feedback from the supervisory unit to further assist with fault detection.

An issue when using feedback from the supervisory unit for fault detection is the chance for a false positive that can prematurely deactivate the redundant components. To help reduce this issue, several techniques have been implemented by the BITs. First, if supervisory feedback has indicated that a fault has occurred within a WMCU, further testing by the BIT can be done to affirm the presence of this fault. This testing can include

communicating with other WMCUs within the device. Second, each WMCU will attempt to confirm that the supervisory unit is not operating erratically by performing a sequence of communication tests. If the supervisory unit fails these tests, the WMCUs can operate without supervisory feedback. Third, if the supervisory unit passes these tests, each WMCU will attempt to communicate with other WMCUs and check whether the supervisory unit is also indicating that they have a similar fault. If the supervisory unit is indicating that a same fault has just occurred, then the supervisory unit is assumed to have failed since near-coincident faults (in which two components fail simultaneously and independently) are typically a rare occurrence. If none of these tests indicate a false positive, then the supervisory feedback must be assumed to be correct.

Upon the detection of a fault, a variety of recovery mechanisms can be implemented, including repeated computations, memory invalidation and soft/hard resets. Should the fault be permanent, then a fail-safe mode is entered to deactivate the corresponding WMCU, as discussed later in Chapter 5.

4.3 Supervisory Diagnostic Algorithm

The introduction of the supervisory unit can help to improve fault coverage by providing additional fault diagnostic capabilities and component feedback. Simply, upon the detection of a mismatched output from either of the WMCU's, the supervisory unit will provide additional information, such as the values sent from each component, back to the individual BITs. A leading issue with software-based BITs is having sufficient time to complete diagnostics to detect a fault since these self-tests must not interfere with normal system operation [40]. Therefore, by notifying each BIT that a fault has occurred, the WMCU's operation can be halted to prevent a system malfunction, allowing for more indepth testing to diagnose the fault. Once the fault is diagnosed, system operation can resume.

A secondary feature of the supervisory unit is that it provides an interfacing point between the two modules. Each WMCU component can fail in a variety of ways that could render its main MCU operational, but the radio inoperable [\(Figure 4.1\)](#page-59-0). By allowing for each module to share resources among one another, an added level of fault-tolerance can be

achieved. This resource sharing is managed by the supervisory unit's diagnostic algorithm, providing a second layer of redundancy into the design. Both features of the supervisory diagnostic algorithm have been shown in [Figure 4.2.](#page-62-0)

Figure 4.2: Features of the supervisory diagnostic algorithm.

Note that the supervisory unit has been designed to be fail-safe wherever possible. If the supervisory unit is working correctly, each module's redundant WMCU will send information to the supervisory unit before that information is sent to its respective radio. If the supervisory unit suffers from a failure, such as a loss of function, its failure will not impact the operation of the WMCUs. Each WMCU can bypass the supervisory unit and send information directly to its respective radio. If the supervisor unit suffers from a malfunction and provides faulty feedback to each WMCU (i.e. a false alarm), this faulty feedback does not necessarily result in a device failure. Each component's BIT is responsible for deciding whether a fault has occurred and uses the supervisory feedback to assist with fault detection.

By having a reasonably fail-safe supervisory unit, ideally only a small fraction of supervisory unit CCF mechanisms (captured by the β_s -factor) have been introduced into the design.

4.4 Supplementary Fault Detection Hardware

To further improve the system's fault coverage, additional fault detection hardware has been developed and introduced into the design. This proprietary hardware is embedded within each digital communication bus. Upon the detection of a fault, electromechanical relays are triggered that isolate each WMCU within the module.

4.5 Design Summary

The proposed redundancy management system consists of several parts. The first is the chosen topology that follows from the derived reliability model in Chapter 3. It utilizes dual-redundancy, diversity and modularization to help improve the reliability. The introduction of the supervisory unit aims to help improve fault coverage by providing feedback to each component's BIT. As well, proprietary fault detection hardware also helps to improve the level of fault coverage. The modularized design and the use of diversity in design have further reduced the impact of CCFs.

Chapter 5

5 WSN Implementation

In this Chapter, the details of the implemented design are presented. The implementation process has been divided into two tasks: implementing the hardware for the prototype WSN devices and developing the software for the WSN devices.

5.1 Hardware Implementation

For the implementation of the hardware, first a diverse set of WSN device components have been selected. Next, key circuit have been simulated using a circuit simulation tool and then the printed circuit boards (PCBs) have been designed.

5.1.1 Diverse Component Selection

The proposed WSN device consists of two dual-redundant and diverse modules, Module A and Module B. Module A operates at 900MHz whereas Module B operates at 500MHz. Each of the WMCUs within a module must be functionally equivalent (i.e. have the same RF modulation scheme) and have a similar failure rate to each other, as described in Section 3.4. As well, each supervisory unit must also be functionality equivalent and share a similar failure rate. The supervisory units should also have a failure rate similar to or lower than that of the WMCUs, as noted in Section 3.1 and Section 3.2.

[Table 5.1](#page-65-0) provides an overview of the components selected for each module. Manufacturer provided failure rates in failure in time (FIT) are shown in the table, if available. The selected components satisfy the diversity requirement for the proposed redundancy management system, as they are designed by different companies and use different controller technology. Further, the supervisory unit's failure rates are similar to that of the WMCUs. This allows for a reliability improvement to be attained by the supervisory units, as detailed in Section 3.1, [Figure 3.4.](#page-42-0)

Module	Controller	Compatibility	Failure Rate
Component			(FIT)
Module A -	Atmel	IEEE 802.15.4	1.22 (90%CL,
ATZB-X0	AVR	BPSK, O-QPSK	55° C)
WMCU			
Module A-	Atmel	IEEE 802.15.4	Not Available
ATSAMR30	$ARM-M0+$	BPSK, O-QPSK	
WMCU			
Module A -	Atmel	CAN 2.0B	1.22 (90%CL,
AT90CAN	AVR		55° C)
Supervisory Unit			
Module B -	Texas	IEEE 802.15.4g	2.41 (60%CL,
CC1310 WMCU	Instruments	GFSK	at 55° C)
	ARM-M3		
Module B -	Silicon	IEEE 802.15.4g	$0.8(60\%CL,$
EZR32LG	Labs	GFSK	at 55° C)
WMCU	ARM-M3		
Module B -	NXP	CAN 2.0B	Not Available
LPC17	ARM-M3		
Supervisory Unit			

Table 5.1: Diverse component selection.

As mentioned in Section 3.4, diversity can improve reliability if non-overlapping failure modes exist between the replicated components. In the proposed design, diversity has been achieved on two fronts. First, two different wireless communication frequency bands have been selected at 900MHz and 500MHz. The advantage of choosing two different bands is the resilience to partial channel blocking (perhaps due to interference in the 900MHz ISM band or due to obstructions). Lower communication frequencies tend to have an improved communication range, potentially allowing the 500MHz radios to maintain a communication link if the 900MHz radios should fail.

Second, a diverse set of controllers have been selected within each module that results in several advantages over a non-diverse set. It is expected that unintentional design flaws with a controller by one manufacturer should not be presented in a different manufacturer's controller. A common example of such a design flaw is the Pentium FDIV bug [41]. As well, different software, programming tools and compilers are needed that can potentially reduce systemic design flaws. Further, it has been determined experimentally that different technology will be impacted differently by harsh environmental conditions [21] [42]. For example, different controller technology will have a different apparent activation energy that results in altered performance at low vs high temperatures. Additionally, different wireless chips are more susceptible to low ionizing radiation dose rates whereas others are more susceptible to high dose rates [42]. Therefore, the selection of diverse components can result in non-overlapping failure mechanisms to reduce the impact of CCFs in a design.

5.1.2 Circuit Simulations

Prior to constructing hardware, key circuit structures have been simulated to verify their design. Three electronic circuit functions have been selected for simulation:

- 1) 4-20mA to 0-2.4V sensor interface.
- 2) RF filtering/matching circuit.
- 3) Fault detection hardware.

Simulations for the sensor interface and the RF circuit have been completed using LTSPICE, a free electronic circuit simulator developed by Linear Technology. The fault detection hardware simulation has been completed using PSPICE, a similar circuit simulator developed by Cadence. The details for these simulations are presented next.

5.1.2.1 Sensor Interface Simulation

The sensor interface is required to convert a 4-20mA industrial sensor signal into a voltage for each MCU's analog to digital converter (ADC). The minimum input voltage for one of the selected MCU's is 2.4V. To prevent component damage, a 20mA signal should yield a voltage of 2.4V to each ADC's input.

An operational amplifier has been selected to convert the sensor's current signal into an appropriately scaled voltage. The schematic for the sensor interface is shown in [Figure 5.1.](#page-67-0) The results of the simulation can be seen in [Figure 5.2.](#page-67-1) Note that a low sampling frequency has been selected that simplifies the sensor interface design since WSNs typically do not require very high sampling frequencies.

Figure 5.2: Sensor interface simulation results.

RF Filtering Simulation $\frac{2}{16}$

Module B's WMCUs are capable of operating across a wide range of sub-GHz radio **R 25 1 C 75 330p C 76 C 77 C 78 L52** frequencies. A RF filter is therefore required to ensure that the radios operates within the 1

500MHz band. Manufacturer suggested RF filter's and matching circuits have been used as the base circuit design and tuned for the 500MHz band. [Figure 5.3](#page-68-0) to [Figure 5.6](#page-69-0) show the schematics and the simulation results for the CC1310 and the EZR32LG WMCU. **20n** range of EXECUTE IN THE ENTITY SUPPORTED IT
WMCUs are capable of operating across a wide range of sub-GHz rad Frequencies. A RF filter is therefore required to ensure that the radios operates within the *RH* filter is there **V 8 0.1p**

Figure 5.3: CC1310 filtering schematic. (Left, TX) (Right, RX).

Figure 5.4: CC1310 filtering simulation. (Left, TX) (Right, RX).

Figure 5.5: EZR32LG filtering schematic. (Left TX) (Right RX).

Figure 5.6: EZR32LG filtering simulation. (Left TX) (Right RX).

For both sets of simulations, the maximum gain occurs at around 580MHz rather than the desired 490MHz center frequency in the selected 500MHz band. It is expected that unmodelled conditions (such as input pin capacitances, trace impedances, and stray capacitances) will center the signal closer to 490MHz. Therefore, during the PCB implementation of Module B, a spectrum analyzer has been used to confirm that that the bandpass signal is shifted closer to 490MHz.

5.1.2.3 Fault Detection Hardware Simulation

In the fault detection hardware simulation, the circuit is expected to detect and identify digital communication bus faults, such as encoding and 'stuck-at' faults. Although the schematic for this hardware has not been shown, one of the simulations results has be presented in [Figure 5.7.](#page-70-0) The results for a 'stuck-at' fault for a digital communication line is shown in the figure. A trip signal is sent after 1 byte of inactivity, or 8 digital pulses (blue line), for a digital line communicating at $100KHz$ using the I²C protocol. The trip signal then triggers a switch that drives current through a relay (red line).

Figure 5.7: Circuit simulation depicting a trip signal for the fault detection hardware.

5.1.3 PCB Modules

The second step in the hardware implementation process is to develop the PCB prototypes for the proposed WSN system. This system consists of two dual-redundant modules, Module A and Module B. Module A has been selected to operate at 900MHz and comprises of the following components: the ATZB-X0 WMCU, the ATSAMR30 WMCU and the AT90CAN supervisory unit.

A dual-redundant 4-20mA sensor interface has also been implemented, along with a dualredundant digital communication bus (SPI and $I²C$) and an external wired CAN bus. Electromechanical relays have been used to reconfigure the device if faults are detected within the various components. The prototype for Module A is depicted in [Figure 5.8.](#page-71-0)

Figure 5.8: Module A PCB prototype.

Module B has been selected to operate at 500MHz and comprises of the following components: the CC1310 WMCU, the EZR32LG WMCU and the LCP17 supervisory unit. The similar features to Module A have been included in Module B, and the prototype is depicted in [Figure 5.9.](#page-71-1)

Figure 5.9: Module B PCB prototype.
A prototype for the modularized device has also been built. In the modular design prototype implementation, each module has been divided into several sub-modules. Module A has been divided into three sub-modules for its two WMCUs and the supervisory unit, A1, A2 and S1 respectively. Similarly, Module B has been divided into three sub-modules for its two WMCUs and supervisory unit, B1, B2 and S2 respectively. These sets of sub-modules are shown in [Figure 5.10](#page-72-0) to [Figure 5.12.](#page-73-0)

Figure 5.10: Sub-modules A1 (left) and A2 (right).

Figure 5.11: Sub-modules B1 (left) and B2 (right).

Figure 5.12: Sub-modules S1 (left) and S2 (right).

The modular design's auxiliary systems (such as the dual-redundant sensor interface and fault reconfiguration mechanisms) have been separated into two sub-modules, AUX1 and AUX 2. The auxiliary sub-modules are shown in [Figure 5.13.](#page-73-1)

Figure 5.13: Sub-modules AUX1 (left) and AUX2 (right).

The proprietary fault detection hardware has also been separated into a sub-module, which is shown in [Figure 5.14.](#page-74-0)

Figure 5.14: Proprietary fault detection hardware sub-module.

The complete modular design is shown in [Figure 5.15.](#page-74-1)

Figure 5.15: Modularized design.

5.2 Software Implementation

Software implementation includes the integration of an operating system (OS) across each embedded platform, the communication stack and the remote server for data-logging.

The complete software stack for the developed WSN system is illustrated in [Figure 5.16.](#page-75-0) At the base-layer are the MCU drivers and radio drivers for controlling the softwarehardware interactions. The second layer from the bottom is the embedded OS that is responsible for scheduling tasks, handling interrupts and managing the application. The third layer is the medium access control (MAC) protocol. The MAC protocol dictates the flow of data to and from a radio to ensure that the medium (wireless link) is accessed through some control scheme. The next layer is the network protocol which is responsible for dictating how data is routed through the network. The top layer is the application and is the location where the various test algorithms reside.

Figure 5.16: Software stack for implementation.

Above the embedded software stack is the Linux remote server driver and the ThingSpeak remote server. The Linux remote server driver interacts with the embedded software through the application layer and is responsible for pushing data to the ThingSpeak server for data logging. Each of these layers is discussed in more depth.

5.2.1 Hardware Drivers

To enable the correct software-hardware interactions on each embedded MCU, a set of hardware drivers is required. [Table 5.2](#page-76-0) identifies the hardware driver that have been developed, the MCUs that required the driver and the functionality provided by the driver.

Driver	Target MCU	Functionality
ADC	EZR32HG, EZR32LG, SAMR30,	Analog to digital conversion for
	ATXMEGA, CC1310, AT90CAN,	the 4-20mA sensor interface
	LPC17, Atmel 8051	
$\mathbf{I}^2\mathbf{C}$	EZR32HG, EZR32LG, SAMR30,	General purpose bus
	ATXMEGA, CC1310, AT90CAN,	communication (multi-master)
	LPC17, Atmel 8051	
SPI	EZR32HG, EZR32LG, SAMR30.	General purpose bus
	ATXMEGA, CC1310, AT90CAN,	communication (multi-master)
	LPC17	
UART	ATXMEGA, SAMR30	Peer-to-peer microcontroller
		interfacing
TIMER	EZR32HG, EZR32LG, SAMR30,	Timer and task scheduling
	ATXMEGA, CC1310, AT90CAN,	
	$LPC17$, Atmel 8051	
CAN 2.0B	AT90CAN, LPC17	CAN bus driver for CAN
		transceiver module

Table 5.2: Developed microcontroller drivers.

Each MCU required drivers for their ADCs, buses $(I²C$ and SPI) and timers. Only the supervisor units required the CAN 2.0B driver. Only the ATXMEGA and the ATSAMR30 adopted the universal asynchronous receiver-transmitter as an added communication channel during early prototyping. These drivers represent the lowest layer of the software stack in [Figure 5.16.](#page-75-0)

5.2.2 Operating System Porting

As noted previously, there are several challenges that arise when working with a diverse set of MCUs. After evaluating several OSs suitable for WSN devices, RIOT OS has been selected [43].

[Table 5.3](#page-77-0) summarizes some of the effort required to port RIOT OS for 8 different MCUs. Note that RIOT OS could not be ported for the uC8051 architecture, since the CPU core's stack in this architecture differs significantly from more modern MCU architectures.

Microcontroller	Architecture Port	Compiler Port	
ATXMEGA	Redefine CPU Registers and CPU Stack Calls. Interrupt Calls	GNU Supported	
CC1310	Direct Support	LTS Compiler Port - remove atomic calls, alter data type definition, change assembly calls	
ATSAMR30	Adaption of the ATSAMR20	GNU Supported	
EZR32HG	Adaptation of the EZR32HG	Arm-gcc Port - minor changes to header support calls, assembly calls	
EZR32LG	Adaptation of the EZR32HG	Arm-gcc Port - minor changes to header support calls, assembly calls	
AT90CAN	Adaption of the Atmega1281	GNU Supported	
LPC17	Direct Support	Arm-gcc Port - minor changes to header support calls, assembly calls	
uC8051	Incompatible - CPU stack does not support	Incompatible - C51 compiler does not support assembly use	

Table 5.3: Summary of porting requirements for RIOT OS.

5.2.3 MAC and Network Layers

The next stack layers in [Figure 5.16](#page-75-0) is the MAC protocol layer and the Network layer. The IEEE 802.15.4 protocol has been chosen as the primary MAC protocol for use due to its wide adoption in industry. Due to the complexity and strict timing of the IEEE 802.15.4 standard, manufacturers provide their own custom implementations for their products. The MAC protocol and its respective drivers are integrated with the Real Time Operating System stack layer through various interfacing points in the software. These interfacing points are the radio driver function calls and callback functions.

A second MAC protocol has been developed for the implementation and used with Module B. The reasoning for this alternate MAC protocol implementation is to demonstrate that the proposed system need not be tied to the IEEE 802.15.4 protocol. By using manufacturer provided radio drivers, the second MAC protocol can be linked into the OS in a similar way described above.

The next layer in the software stack is the network protocol. The network protocol is responsible for routing data from end to end in the network. Since the network protocol is not the primary focus of this work, a static routing table has been used for fixed path data routing.

5.2.4 Application Layer

The application layer of the software distinguishes devices from one another. A device either acts as a field device, a router or a gateway in the network. Field devices poll their ADC for sensor data, format the information into a packet and then send this packet towards the gateway device. Router devices act as intermediate devices between the field devices and the gateway. These devices only re-transmit received data in the direction of the gateway. The gateway device acts as the network sink, aggregating the received data and passing it to an external network. In this application, the external network is the ThingSpeak remote server. The gateway devices communicate externally to a computer that then connects to the remote server through an Internet connection. Note that the application layer also houses the microcontroller-based BIT and the supervisory diagnostic algorithm.

5.2.5 Remote Server Integration

The final software layers are the Linux drivers and the ThingSpeak remote server. ThingSpeak is a free platform for IoT data collection, data processing and action control. Using the HTTP, devices with an Internet connection can push and poll data to/from the ThinkSpeak's server. The purpose of the server in this application is to demonstrate the capabilities for the proposed system to interface with an external network and log data.

A computer acts as an intermediate device between a gateway's application layer and the ThingSpeak server. A Python script on the computer acts as the ThingSpeak driver, formatting the collected data and then pushing it the ThingSpeak server over HTTP.

5.3 Implementation Summary

The implementation phase consisted of two core tasks. First, the hardware for the WSN has been implemented. This process included the diverse component selection, circuit simulations and the PCB design. The selection of the diverse components and the supervisory unit satisfied the failure rate requirements from the analysis in Chapter 3. After, the software for the WSN has been implemented to allow for the devices to operate as a WSN system. The software implementation included peripheral drivers, an OS, MAC protocols, a network protocol, and gateway interfacing capabilities to the ThingSpeak remote server.

Chapter 6

6 Redundancy Management System Evaluation

The evaluation of the prototype WSN device has been done using a three-fold process. First, to better understand the reliability improvement gained through the proposed redundancy management system, a comparative analysis is performed against existing commercial WSN products. Next, fault injection testing is used to demonstrate the proposed system's fault-tolerance. This testing shows how fault coverage is improved by the supervisory unit and fault detection hardware, while also showing how CCFs can be reduced through the modularized and diverse system. Finally, several experimental test cases are used to demonstrate the WSN's ability to perform industrial monitoring in an experimental setting. These evaluations have demonstrated the effectiveness of the proposed redundancy management system.

6.1 Reliability Analysis

The first evaluation method for the redundancy management system is reliability analysis. Using the analysis from Chapter 2 and Chapter 3, a model can be developed for a onemodule device and a two-module device. For this evaluation, averaged manufacturer failure rate data has been used to represent the component failure rates. Two scaling factors modify the failure rates for harsh industrial environments. First, MIL-HDBK-217 has scaled the component failure rates to an ambient temperature of 105° C and 165° C. These temperatures are selected based on estimated temperature in a NPP during an accident condition [44]. The second scaling term is for the radiation degradation factor that results under 10Krads of ionizing radiation, chosen based on the same previous accident scenario. Note that 10Krads has been chosen since some WMCUs can only withstand about 20- 30Krads TID [45]. The environmental conditions from [44] are shown in [Table 6.1.](#page-81-0) A summary of the failure rates used for the various components are shown in [Table 6.2.](#page-81-1)

Operating Environment	Temperature °C (Ambient)	Radiation Krad (TID)
Control Building (N)	15-40	< 0.2
Auxiliary Building (N)	1-40	$0.01 - 1000$
Auxiliary Building (A)	40-160	$0.01 - 1000$
Loop Compartment (N)	15-40	6000
Loop Compartment (A)	$120 - 200$	8000

Table 6.1: Environment within a NPP during normal (N) and accident (A) conditions.

Table 6.2: Failure rates and scaling factors for various components.

Component	Failure Rate Estimates	Scaling Factor AF $(105^{\circ}C)$	Scaling Factor AF $(165^{\circ}C)$	Radiation Degradation Factor Δ_k
	(FIT)			(10Krad TID)
LiPo Battery*	10000	Not Available	Not Available	Not Available
Power Converter*	20000	5.0	47.7	0.1408
Wireless Microcontroller**	1.48	5.0	47.7	0.1026
Supervisory Microcontroller**	1.22	5.0	47.7	0.0638
Sensor Interface*	20000	5.0	47.7	0.2377
RF Circuitry*	5000	27.6	64.4	0
Digital Bus*	20000	5.0	47.7	0.133

*FIT estimated using MIL-HDBK-217F **FIT derived from chip manufacturer

To evaluate the performance of the developed devices, a comparative analysis against two existing WSN platforms, the IRIS mote [46] and the Meshlium Gateway [47], has been performed. These two systems are assumed to contain the same major device components and failure rates as the proposed WSN device. Note that the IRIS mote is a simple, nonredundant system whereas the Meshlium gateway incorporates redundancy. As well, it has been assumed that a BIT-based redundancy management system exists within the Meshlium gateway that does not have the proposed supervisory unit or fault detection hardware. [Figure 6.1](#page-82-0) shows these two wireless devices.

Figure 6.1: IRIS mote (left) and the Meshlium gateway (right) device.

The final parameters required for reliability modelling is the fault coverage, c , and the CCF's β -factor at each layer of the design. Without experimental data, accurate estimates for these parameters can be challenging. Rather than arbitrarily selecting values, an optimistic evaluation of the proposed system has been done. Here, the fault coverage level, c, is assumed to be one and the β -factors are assumed to be zero for each of the analyzed devices, when applicable. That is to say, the system has perfect fault coverage and no CCF mechanisms introduced into the design. Although these model parameters are unrealistic, all three systems are being evaluated under the same assumed conditions.

Under these assumptions for the model parameters, [Figure 6.2](#page-83-0) (for 105° C) and [Figure 6.3](#page-83-1) (for 165° C) show the reliability for four devices: the IRIS mote, the Meshlium gateway, a one-module device, and the two-module device. Note that for the latter two, each module is a dual-redundant design by itself, with its own supervisor.

Figure 6.2: Reliability comparison for different devices under elevated environmental conditions (105^oC).

Figure 6.3: Reliability comparison for different devices under elevated environmental conditions (165^oC).

From these figures, it is clear that both one-module and two-module devices can provide a MTTF improvement over the commercially available ones. This result is anticipated since the proposed system incorporates redundancy within each module and between the two modules. Nevertheless, this result shows how the proposed topology can improve system reliability, making it better suited for deployment in harsh environments. Furthermore, these two figures also clearly demonstrate the impact of harsh environment on system reliability.

6.2 Fault Injection Testing

The second method for evaluating the developed WSN devices is fault injection testing [48]. Here, a set of controlled tests are performed on both one-module and two-module devices to determine whether the systems can tolerate the faults. In these tests, faults are either emulated within each module through software or are physically injected. [Figure 6.4](#page-84-0) illustrate these two fault injection methods.

Figure 6.4: Method for injecting faults into a device.

Multiple fault injection tests have been performed on Module A, Module B and their combination as a two-module device. Dummy sensor data has been used as module inputs,

which is then transmitted through any of the module's radios, as shown in [Figure 6.5.](#page-85-0) Depending on the specific faults injected, it is expected that a one-module device will fail, whereas the two-module device should operate successfully.

Figure 6.5: Fault injection test scenario.

Two rounds of fault injection testing have been performed, the first to investigate the fault coverage improvement, and the second is to evaluate the CCF reduction. For the first round, the injected faults and the detection location (either the BIT, the supervisory diagnostic algorithm or the fault detection hardware) are shown in [Table 6.3.](#page-87-0)

As shown in [Table 6.3,](#page-87-0) the microcontroller BIT can detect four of the injected faults. Faults, such as out-of-bounds outputs (test #1.4) and deadlock/livelock situations (test #1.2 and test #1.3 respectively) can also be detected by standard software-based BITs. However, certain faults can only be detected by the supervisory diagnostic algorithm or the fault detection hardware, which has demonstrated the superiority of the proposed design. As shown in the table, in test #1.5, digital line noise has been injected into one of the device's redundant WMCUs that caused its output value to be changed. The supervisory diagnostic algorithm has detected a mismatched output between the dual-redundant WMCUs and has provided feedback to both microcontrollers. Each microcontroller has then compared their intended output value with the received one and has identified that a fault has been induced by digital line noise. A similar process has occurred in test #1.6, and the supervisory diagnostic algorithm has been able to help detect the fault successfully.

The fault detection hardware has also contributed to the improvement of fault detection in a device. The stuck-at faults in test #1.8 and test #1.9 caused a digital communication line to become unusable. Within a single module, three components share each digital communication line, preventing BITs from identifying the source of the fault (any component could have caused the fault). The additional fault detection hardware has been designed to identify the source of these types of digital communication faults, and therefore contributes to an improved fault coverage.

Note that the proposed WSN device can still suffer from external faults. For example, if the sensor inputs faulty data (test #1.5), it cannot be detected by the proposed system. This can be handled separately, but handling sensor faults are out of the scope of this work.

In a non-redundant WSN devices (such as the IRIS mote in the previous analysis), it is reasonable to assume that the faults in [Table 6.3](#page-87-0) could not be detected and recovered from as they do not employ a redundancy management system. To contrast, existing redundant WSN devices (such as the Meshlium gateway) might be able to detect faults that are detected by the microcontroller BIT. However, standard redundant systems that use BITs cannot detect and recover from the faults that are covered only by the supervisory diagnostic algorithm or by the fault detection hardware (see [Table 6.3\)](#page-87-0).

These tests have demonstrated that additional faults can be detected, and hence, a device's fault coverage can be improved by the proposed system. To determine the exact fault coverage improvement, exhaustive testing has to be done, which is beyond the scope of this work.

Test	Faults Injected	Location Detected			
#		Microcontroller	Supervisory	Fault	Undetected
		BIT	Diagnostic	Detection	
			Algorithm	Hardware	
1.1	WMCU Memory Corruption	X			
1.2	Deadlock	X			
1.3	Livelock	X			
1.4	Out-of-Bounds Output	X			
1.5	Sensor Input Fault				X
1.6	Digital Line Noise		X		
1.7	Latched Digital Output Register		X		
1.8	Digital Line 'Stuck-at' Fault Low			X	
1.9	Digital Line 'Stuck-at' Fault High			X	
1.10	Digital Line Encoding Fault			X	
	Totals		2	3	

Table 6.3: Results of the fault coverage fault injection tests.

The second round of fault injection testing has been done to evaluate the impact of CCF mechanisms in the proposed WSN device. To demonstrate that, faults have been injected into one-module devices and a two-module device, and the fault-tolerance performance has been compared, as shown in [Table 6.4.](#page-88-0) Faults have been injected into each module and into the redundant/diverse sub-systems (i.e., the dual-redundant bus and dual-redundant sensor interface). In [Table 6.4,](#page-88-0) WMCU A1 Fault represents a complete failure of the module A's first redundant wireless microcontroller. If the device is able to complete the test scenario previously described (acquire dummy data and then send the data through a radio), then the result of that test is a success and is shown as a 'pass' in the table. Conversely, if the device is unable to complete the test scenario, then the result of that test is shown as a 'fail'. Note that in the table, some tests may not be applicable for certain devices, which are indicated as N/A.

It can be seen from [Table 6.4](#page-88-0) that the redundant design has reduced certain single points of failures in the device. Each device has a redundant sensor interface and a redundant digital bus. A single fault in either of these (test #2.16 to test #2.19) have not caused the one-module devices or the two-module device to fail.

Test	Faults Injected	Results (Pass / Fail)		
#				
		Module A	Module B	Module $A+B$
		(One-module)	(One-module)	(Two-module)
2.1	WMCU A1 Fault	PASS	N/A	PASS
2.2	WMCU A2 Fault	PASS	N/A	PASS
2.3	WMCU B1 Fault	N/A	PASS	PASS
2.4	WMCU B2 Fault	N/A	PASS	PASS
2.5	WMCU A1+A2 Fault	FAIL	N/A	PASS
2.6	WMCU B1+B2 Fault	N/A	FAIL	PASS
2.7	WMCU A1+A2+B1 Fault	N/A	N/A	PASS
2.8	WMCU A1+A2+B2 Fault	N/A	N/A	PASS
2.9	WMCU B1+B2+A1 Fault	N/A	N/A	PASS
2.10	WMCU B1+B2+A2 Fault	N/A	N/A	PASS
2.11	Supervisory S1 Fault	PASS	N/A	PASS
2.12	Supervisory S2 Fault	N/A	PASS	PASS
2.13	Supervisory $S1+S2$ Fault	N/A	N/A	PASS
2.14	900MHz Channel Blocking	FAIL	PASS	PASS
2.15	500MHz Channel Blocking	PASS	FAIL	PASS
2.16	Sensor Interface S1 Fault	PASS	PASS	PASS
2.17	Sensor Interface S2 Fault	PASS	PASS	PASS
2.18	Digital Bus D1 Fault	PASS	PASS	PASS
2.19	Digital Bus D2 Fault	PASS	PASS	PASS
	Totals (Passed)	8	8	19

Table 6.4: Results of the CCF fault injection tests.

In certain scenarios, the effect of CCFs have only been mitigated by the diverse, modular design. For example, in test #2.1 and test #2.2, module A (and therefore module A+B) has continued to work when any one of module A's WMCU has suffered from a fault (this is expected since each module has dual-redundant WMCUs). However, when both WMCUs within a single module have failed (test #2.5 and test #2.6) only the two-module device has continued to work. This is because, as designed, the modular device can work if any one of its modules are operational. Further, the two-module device has continued to work when only one WMCU (from four WMCUs from module A and B) has been operational (test # 2.7 to test # 2.10). Moreover, partial channel blocking, such as the 900MHz channel blocking and the 500MHz channel blocking in test #2.14 and test #2.15 respectively, has been mitigated by the diverse, two-module device.

The fail-safe nature of the supervisory unit has also been demonstrated. The one-module and two-module devices has continued to operate even when the supervisory unit has suffered from a loss of function fault (test #2.11 and test #2.12).

In summary, in the proposed design, each dual-redundant module has a supervisory unit, fault detection hardware and reconfiguration mechanisms. If a CCF mechanism causes one module to fail, the second module in the modularized design does not necessarily fail since the redundancy management system has also been modularized. Existing WSN devices might not have this partially-modularized redundancy management system. Therefore, it would be expected that any failure in their redundancy management system would constitute a device failure.

6.3 WSN Experimental Test Scenarios

The third evaluation method verifies that the developed devices can effectively perform industrial WSN monitoring tasks. To do this, the Nuclear Plant Control Test Facility (NPCTF) has been selected as the test platform. The NPCTF is a physical system that emulates the main process loops of a CANDU-style nuclear power plant. As such, this system has an abundant of process variables that can be accessed for monitoring and control purposes. [Figure 6.6](#page-89-0) shows the NPCTF system. For this evaluation, the main loop pressure, denoted as P1, has been selected for monitoring.

Figure 6.6: Nuclear Plant Control Test Facility.

Two test scenarios have been selected. The first test has utilized one-module devices to measure P1, and then relay this information to a gateway device connected to a remote server. The second test scenario is similar, except that two-module devices have been used.

6.3.1 Test Scenario #1: One-Module Data Trending

The first test scenario demonstrates that a one-module device can be used to create a WSN system that can collect industrial process data. By relaying information from a field device to a gateway device, the developed devices can clearly operate together to form a WSN system. [Figure 6.7](#page-90-0) and [Figure 6.8](#page-90-1) show the experimental setup for the three devices using the NPCTF system.

Figure 6.7: A one-module device interfaced to the NPCTF.

Figure 6.8: Test scenario #1 setup.

In this test, the field device has been programmed to poll the sensor interface on 20 second intervals for approximately 37 minutes. After each interval, the P1 value has been forwarded to the gateway device and then uploaded to the ThingSpeak server. The results for this test are shown in [Figure 6.9](#page-91-0) and [Table 6.5.](#page-91-1)

Figure 6.9: Test scenario #1 ThingSpeak server results.

The results of the test show that the one-module devices achieved an event loss rate of 1.81% and was successfully able to trend the recorded pressure sensor data. The success of this first test affirms that a one-module device can perform a general industrial WSN monitoring task.

Table 6.5: Event loss rate results for test scenario #1.

Events Sent	Events Received	Events Lost	Event Loss Rate
	10		1.81%

6.3.2 Test Scenario #2: Two-Module Data Trending

The second test scenario uses a two-module device to complete the described monitoring task. The interfacing of the modules together is shown in [Figure 6.10.](#page-92-0)

Figure 6.10: Interfacing of the two-module device.

The experimental setup using the NPCTF is shown in [Figure 6.11.](#page-92-1) Similar to the previous test scenario, these devices have been programmed to relay process information to the gateway devices on 20 second intervals. This test ran for approximately 6 hours over a three-day span. The results of this test are shown in [Figure 6.12](#page-93-0) and [Table 6.6.](#page-93-1)

Figure 6.11: Test scenario #2 setup.

Figure 6.12: Test scenario #2 ThingSpeak server results.

The results of the second test scenario show an improvement in the event loss rate for the two-module devices to 0.09%. This improvement is expected since both the 900MHz module and the 500MHz module can operate simultaneously.

Events Sent	Events Received	Events Lost	Event Loss Rate
1093	1092		0.09%

Table 6.6: Event loss rate results for test #2.

Both test scenarios have been successful; information has been relayed from the field device to the gateway device, and then uploaded to the remote server. This success concludes the demonstration of the WSN as a general industrial monitoring solution.

6.4 Evaluation Summary

The results of the three evaluation methods have shown the fault-tolerant capabilities of the proposed redundancy management system. First, the comparative reliability analysis of the proposed WSN system has shown to increase reliability against existing commercial devices. Next, the improvement to fault coverage and the reduction in CCF mechanisms has been shown by the fault injection tests. Finally, the WSN's ability to complete a general industrial monitoring tasks has shown that the proposed system can meet the technology's

core requirements. Overall, the evaluation has demonstrated the proposed redundancy management system's ability to improve fault coverage and reduce CCFs in a WSN device.

Chapter 7

7 Conclusions

In this chapter, this work has been summarized and concluded, along with a list of contributions.

7.1 Summary

To understand the reliability performance of redundant systems using BITs, first, a redundancy-relevance boundary has been developed that can be used to identify the minimum fault coverage level to justify a redundant design. This boundary has then been extended to include the impact of CCFs. A reliability-improvement plane has also been developed to determine the reliability improvement gained from a specific design. The redundancy-relevance boundary and the reliability-improvement plane have been shown to be a useful tool when analyzing whether a redundant design can potentially improve a system's reliability.

Next, a redundancy management system topology has been proposed that uses a supervisory unit to improve fault detection. By improving the fault detection capabilities for each BIT, an overall fault coverage improvement can be achieved. This topology has been extended to form a modularized system design that can help to alleviate the impact of CCFs. A redundancy management system has then been designed based on the proposed topology and implemented in a prototype WSN system.

Through reliability analysis and fault injection testing, it has been shown that the proposed system can effectively alleviate the impact of both imperfect fault coverage and CCFs. The suitability of the proposed WSN system under an industrial deployment has also been demonstrated. In summary, the proposed design can be implemented in fault-tolerant WSN devices to make WSNs more robust to withstand harsh environments.

7.2 Contributions

The contributions are listed as follows:

- A reliability-relevance boundary and a reliability-improvement plane has been developed as a tool to aid in assessing the reliability of a redundant design.
- A redundancy management system topology has been proposed that uses a supervisory unit to improve fault coverage in a BIT-based design.
- The proposed redundancy management system topology has been extended to form a modularized system design that can help to alleviate the impact of CCFs.
- Fault detection hardware has been developed for digital communication buses, such as $I²C$, that can improve fault coverage.
- The proposed design has been implemented in prototype WSN devices, and the fault-tolerant performance of the devices has been evaluated.

7.3 Conclusions

A WSN system, when deployed under harsh environment to accomplish a mission, may experience higher rate of component failure, leading the system to fail prematurely. Faulttolerant design based on redundancy can enhance the performance of a WSN system under such deployments. However, overall system performance improvement can be compromised due to factors such as imperfect fault coverage and CCFs. The BIT-based approach for redundancy management suffers from both factors. A BIT-based redundant WSN system has been designed, developed and investigated that makes use of a supervisory unit and a modular architecture to address the issues associated with imperfect fault coverage and CCFs. Based on the evaluation results, it may be concluded that the combination of a supervisory unit and modular design can potentially alleviate the impact of both imperfect fault coverage and CCFs on a redundant WSN design, which may to lead to higher system reliability and fault-tolerance.

7.4 Suggestions for Future Work

To improve upon the existing work, three suggestions are provided to help guide future work on the proposed redundancy management system:

- Evaluate the reliability model developed in Chapter 3 for the supervisory unit using advanced modelling or experimental testing.
- Complete exhaustive fault injection testing to determine the supervisory unit's fault coverage improvement.
- Evaluate the impact of false-positives caused by the redundancy management system on a device's reliability.

The reliability model in Chapter 3 (Equation 3.14) used the coverage decay function to bound the reliability of the proposed redundancy management system. It is not known whether this model accurately reflects the reliability of the system. However, its accuracy could be determined by using more advanced modelling techniques (such as dynamic fault tree analysis) or by completing experimental testing.

It has been shown that the proposed redundancy management system can improve fault coverage, but the magnitude of the improvement is not yet known. Ultimately, the value of the redundancy management system hinges on the fault coverage improvement.

Throughout the analysis of the redundancy management system, it has been assumed that false-positives provided by the supervisory unit can be correctly discerned by each redundant component's BIT. Since the supervisory unit is only providing feedback to each BIT in the proposed design, this assumption might be valid. However, more responsibilities and decision making could be given to the supervisory unit to further improve fault coverage. If this is the case, false-positives could be a prominent issue. Further, each BIT is also susceptible to false-positives. The impact of false-positives from all elements of the proposed redundancy management should be investigated to conclude how effective the design is at improving reliability.

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